# **Mechanical Properties of Cantilevers**

the AFM probe consists of an celtra-sharp tip mounted on the flexible beam (countilever). The role of the countilever is to translate the force active on the tip into or measurable deflection.

It turns out, that the force and doct contilever deflection are related through a rruple linear relationship

F = K DZ

where k is so a spring courtaint of a courtlever.

In this lecture we will focus on deriving this relationship and on determining, how k depends on such properties of a contilever or its size, shape and

# **Center of Mass**

system of N particles

豆=十克miri M=克m:

m: - man of an individual

r: - particle position

Beelk body

E = IN Ill grav

g(r) - deuxity

M - mass of

(The integrals com be extended over all opene, once good outside the locally)

tem

it density is centrom

The motion of the center of morn is defermined by the linear momentum theorem:

M R = F

P - total externel torce

### **Rotational Motion**

- Moment of force (torque) with respect to point 0

No = P x = (No = r = sin x)

- Angular momentum
Lo = P x B = m (P x 3)

- Moment of inertia with respect to any axis (e.g. z)

- N-ponticles

hadins of gyration Kz

M K2 = I2

Moment of mertia describes the distribution of mass within the body and is a rotational analog of man

# Moment of Inertia as Rotational Analog of Mass

Consider kinetic energy T. In rectilinear motion

T= 2 m 02

In notational metron

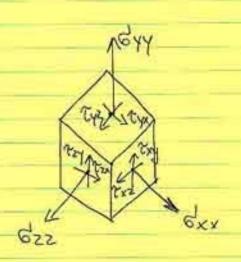
T = 1 I w2

show for a term ring ri couristing of individual particles mi each

for an individual particle  $T_i = m_i \frac{Q_i^2}{2} = \frac{m_i}{2} \left(\frac{2\Pi T_i}{T}\right)^2$   $\frac{2\Pi}{T} = \omega$   $T_i = \frac{m_i}{2} \frac{\omega^2 \Gamma_i^2}{2}$   $T_{total} = \frac{Q^2}{2} \sum_{i=1}^{N} m_i \Gamma_i^2 =$ 

 $-\frac{1}{2}I_2\omega^2$ 

## **Stress and Strain**



Consider a force F acting on a three-dimensional body in a rectangular coordinate nymem

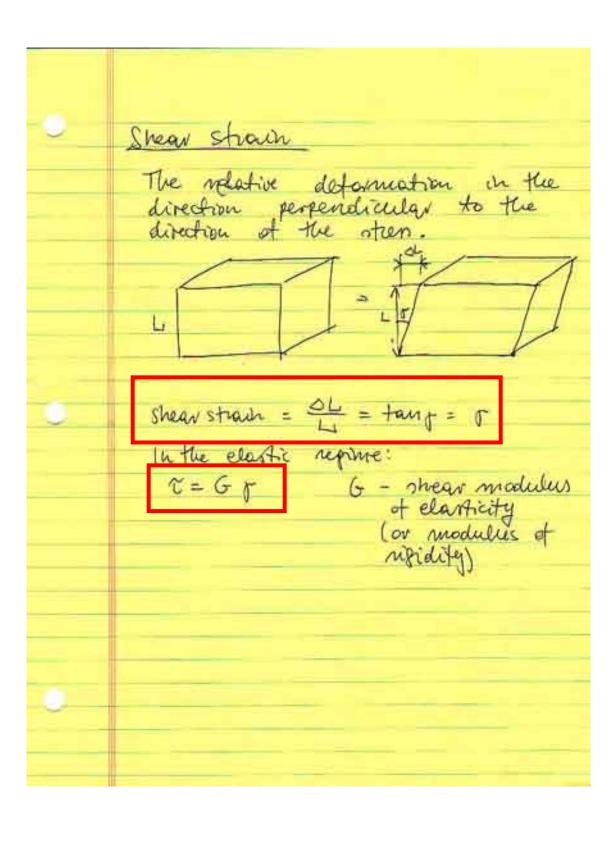
For an element of area Ax perpendicula to the x direction one can identity three stress components:

normal stress: 6xx=lim Fx
(axial) Ax

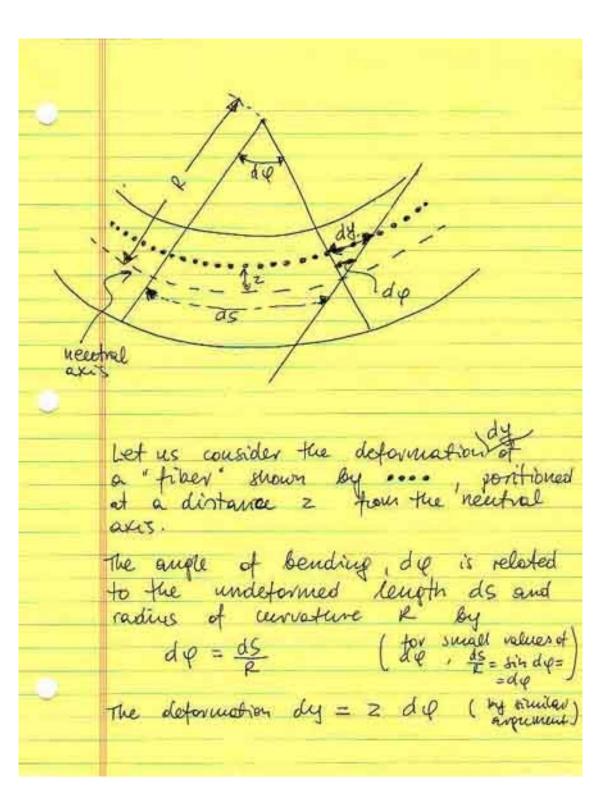
shear strenes: txy = lin Fy
Ax>0 Ax

(Stress components for) TXZ = lim Fx
Ax
Axiables allowed

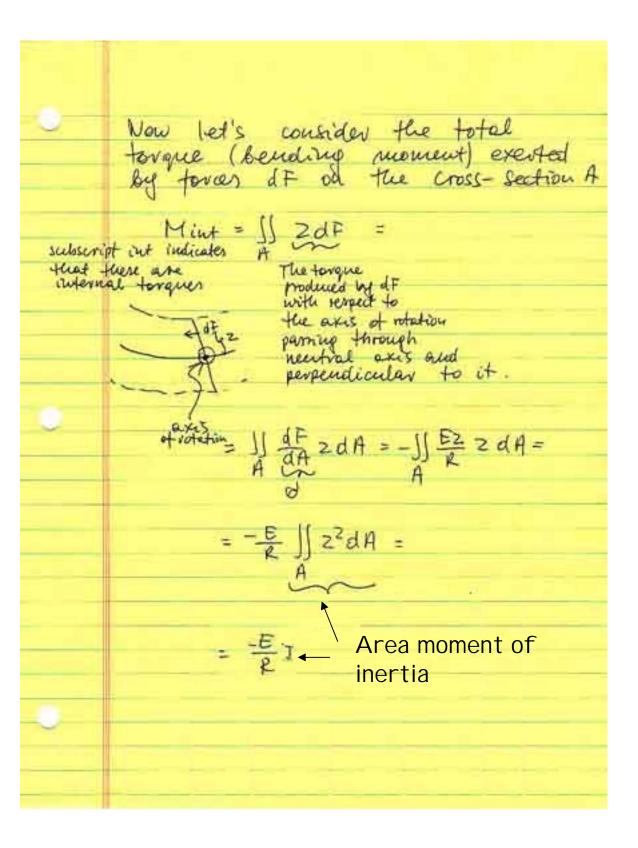
Thus, a vectorial field of torce (Fx, Fy, Fz) gives rise to a tensorial field of stress /dex Cxy exz Czx Czy dzz) In response to troo types of strens (exial and shear), the body will experience a deformation that can be described by two types of strain ( too simplicity we drop tensorial motation): Axial strain Describes the relative elongation be of a bar of length el ist 6 = de 2+50 Within elartic regime the aixid stren and strain are telated by , where E - Young's modulus 6= Ee



# **Balance of Moments in Bending Beam** Consider a beam bending around the x ours perpendicular to the nerfore of this pape: We realize that bending leads to compression and dilation in parts of a beam positioned respectively above and below the neutral line passing through the center of the beam.



Thus the strain in the "fiber" is equal to E = - dy = -2 The corresponding stren 6 = dF = EE = - E= where E is the Young's modulus of the bearn. The total axial force acting on the cross-section perpendicular to the neutral exis is equal to Fa = | dF = - \ R A ZdA since upon bending there is no net change in length of the beam, Fa = 0 This implies that the wenter layer contains the center of mass of the cross-sectional area of the beam ( why?) ( see the definition of the



It will become clear in a moment that it is metal to rewrite that as

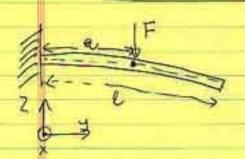
as R = - Mint E I

In equilibrium the internal torques one in balance with the external torque measured with respect to the same axis

Mint + Mext = 0

Tuus

R = Mect



In order to proceed further, we use the expression for the universe of a plane were z (y)

$$\frac{1}{R} = \frac{d^2z/dy^2}{\left[1 + \left[\frac{dz}{dy}\right]^2\right]^{3/2}}$$

notice for small aerosteres is

1 = d2z ,

and thus we obtain the differential aquation describing the balance of torques

d22 = Mect

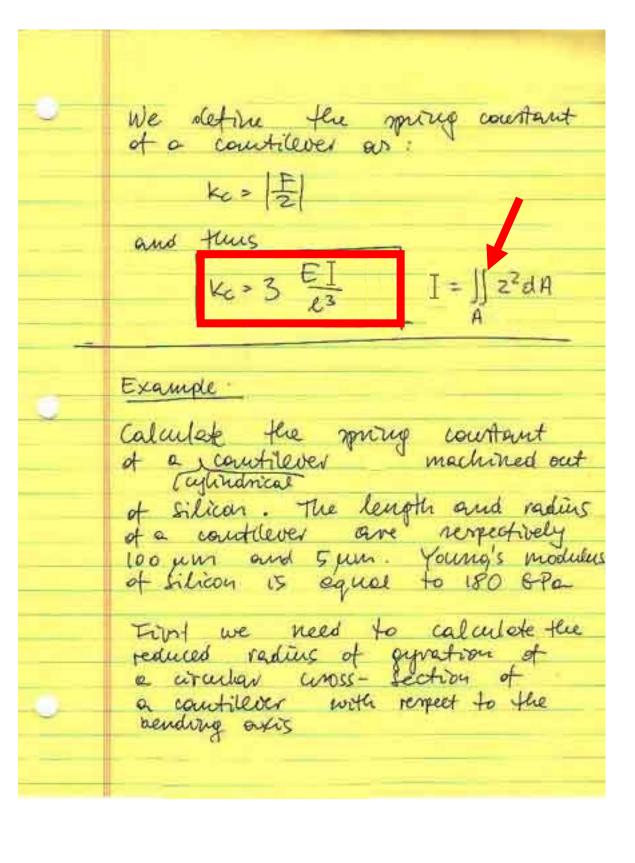
The external torque produced by the torce F acting at a distance "a" from the point of support of the contilever, calculated noith respect to any rotation axis positioned at a distance y < a from the point of support

Mext = F (y-a),

Thus we have to notice

d22 = F (y-a).

The first integration gives  $\frac{dz}{dy} = \frac{1}{2} \frac{F}{FI} y^2 - \frac{F}{FI} ay + C,$ how the boundary condition  $\frac{dz}{dy} = 0 \quad \text{for } y=0$ we obtain C1 =0. The second extegnation gives z = \frac{Fy^2}{6 \in I} (4-3a) + Cz. Again, from the boundary condition 2(4) =0 for 4=0 > C2 =0 Thus, if the force F is applied to the free end of the contilever (a = e), the 2 position of its per end (y = e) is given by  $2 = -\frac{\ell^3}{3 \in I} \cdot F$ 



Now we can use the perpendicular axis theorem, solvide states that: For a planar object the moment of mestic about our axis perpundicular to the plane is equal to the new of the plane is moments of chertia two perpendicular axes passing through the some point in the plane of the object simple justification OIx= OM X2 -om DIY= smy2 DIX+DIY = = DM (x2+4)=  $I_z = I_x + I_y$ = om r2=012 the moment we are seeling I = JT &4

Thus, Kc=3.180 ×109 N . JT. (5 × 10-6 m)4 \$ 220 N